

The Technology Issues of Providing PNT from LEO Satellite Systems

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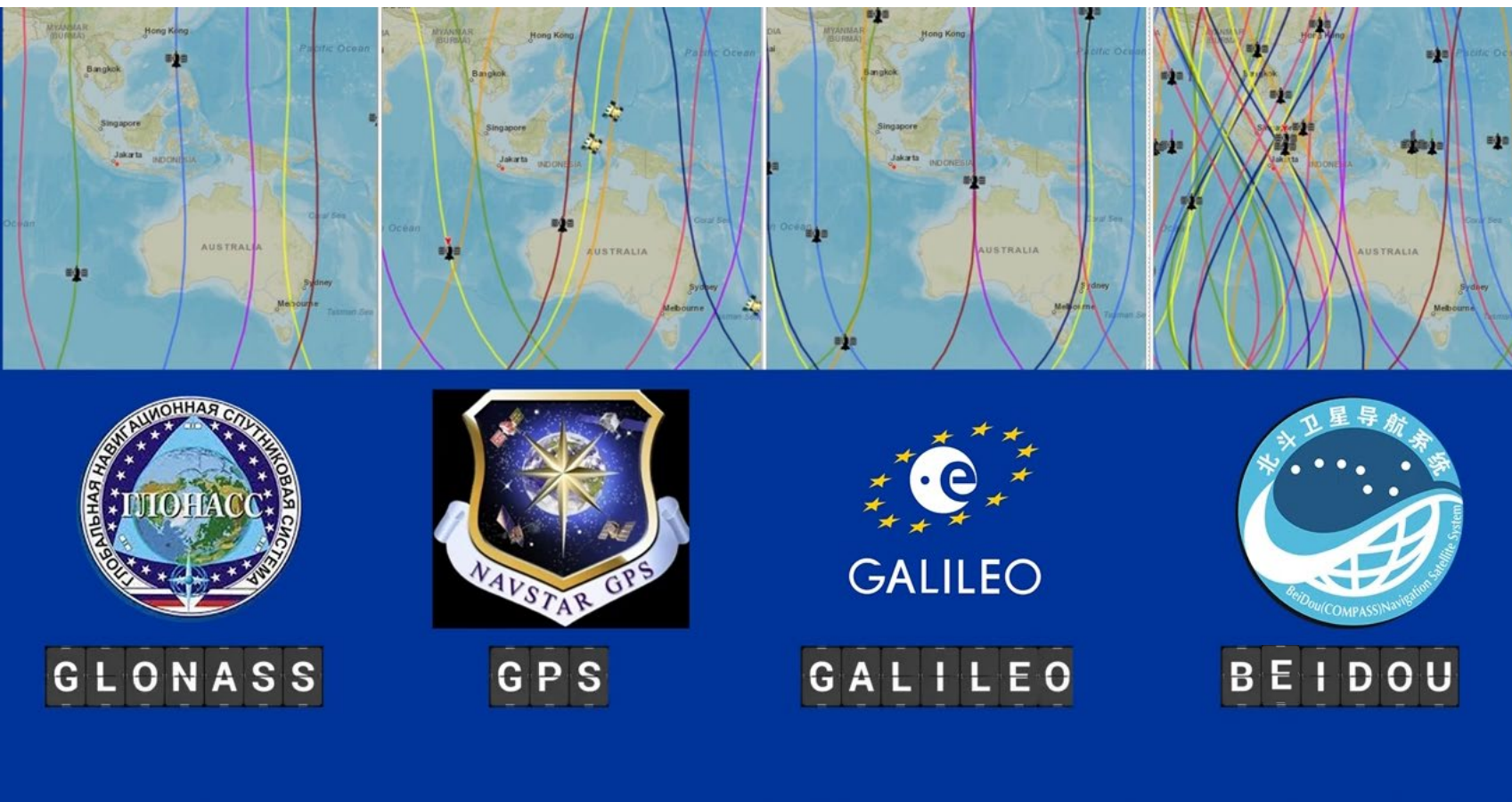
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GPS and GNSS are Amazing



Communication Systems



Electrical Power Grids



Financial Networks



Internet of Things



Navigation Systems



Transportation & Logistics



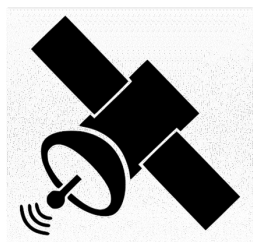
Location-Based Authentication



Maritime

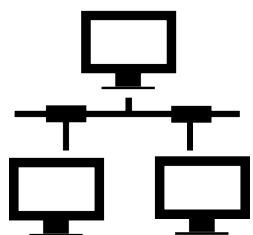


Multiple Technologies Can Provide GPS Resilience



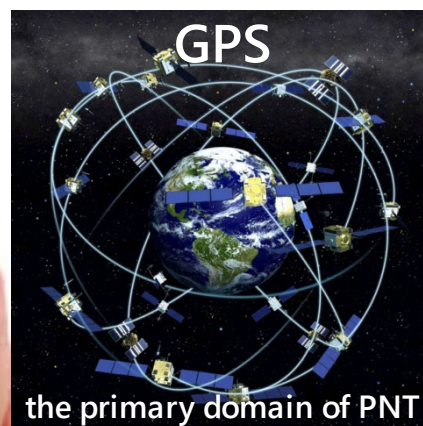
Low-Earth Orbit (LEO) Satellites

Timing and location from an orbit in space
about 25x closer than GNSS



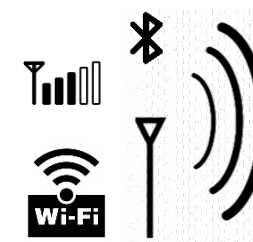
Network Time Transfer

Precise timing from synchronized clocks
across a high-speed computer network



Terrestrial Wireless Infrastructure

Timing and location from ground-based
equipment and support operations across
a specific geographic region



Signals of Opportunity

Location information derived from radio
signals not intended for navigation



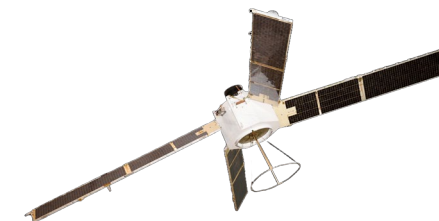
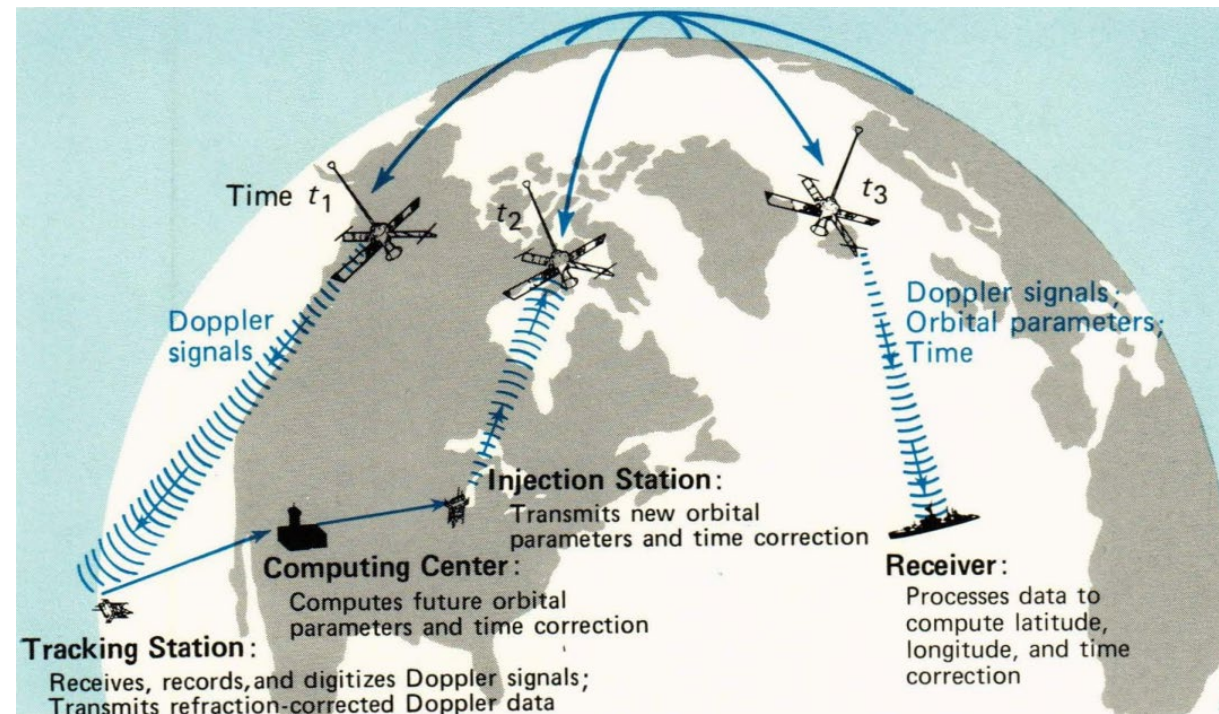
LEO PNT is NOT New

Transit – the Navy Navigation Satellite System (NNSS) – was developed by JHU/APL and deployed by the U.S. DoD in the late 1950s and early 1960s as the first LEO PNT system.

- Fully operational capability: 1964
- Constellation size at FOC: 36 satellites
- Orbit type: Polar
- Orbital altitude: 690 miles (1,100 km)
- Orbital period: 106 minutes
- Frequencies: 150 MHz and 400 MHz
- Accuracy: 200 meters / 50 microseconds

The system used Doppler rather than ranging measurements.

The Transit constellation was obsoleted as a PNT system by GPS in 1996 (although some SVs continued to operate in an ionospheric monitoring mission).





GPS – Initial Operational Capability in 1993

Key new distinctions of GPS over Transit:

- Included signal time of arrival – not just Doppler – allowing it to be much more accurate
- 24x7 global coverage
- Atomic clocks on the satellites for high accuracy
- L-band signal (1.2/1.6 GHz)

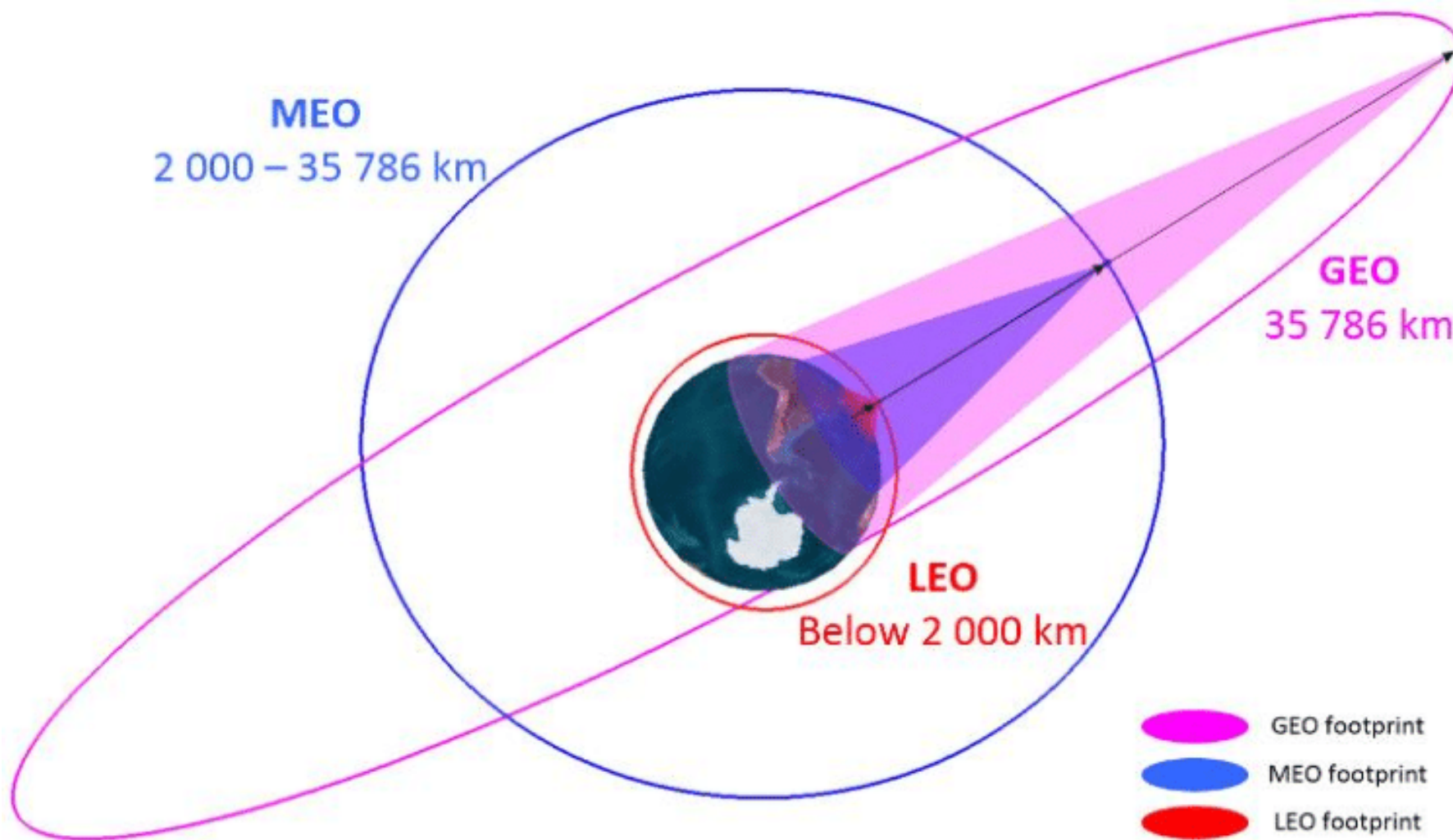
- MEO, not LEO

Why did they do this?





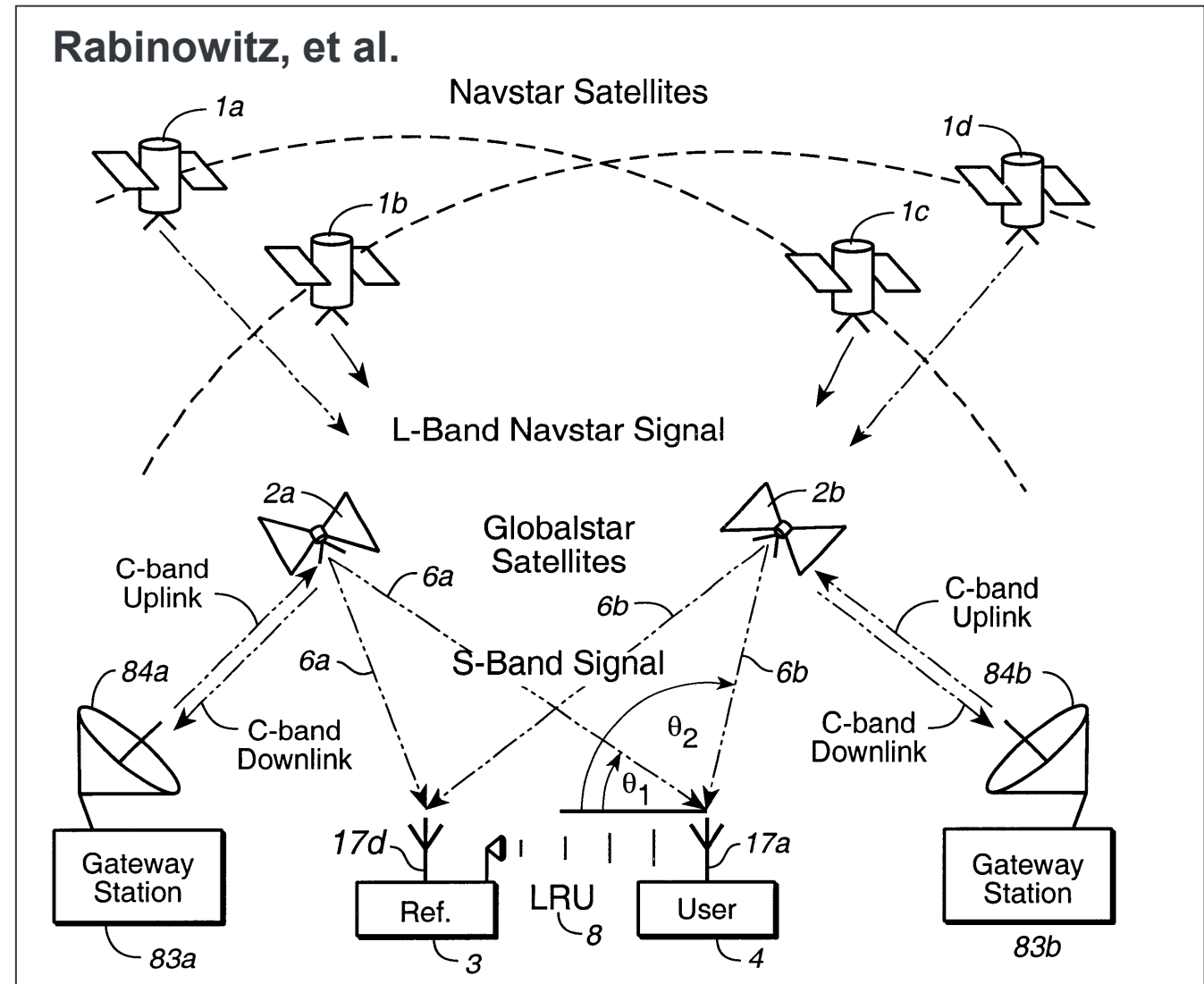
A Key System Trade-off – Orbit Regime





The Re-emergence of LEO PNT

- Commercial SATCOM constellations became operational in the late 1990s
- Researchers began evaluating PNT uses in 2000
- LEO PNT from SATCOM was demonstrated at Stanford





Iridium® Satellite Network

The only satellite network covering the entire globe

66 Active Satellites

- 6 orbital planes of 11 satellites each
- Plus 14 in-orbit spares

Low-Earth Orbit (LEO)

- Low latency
- High power
- Smaller units

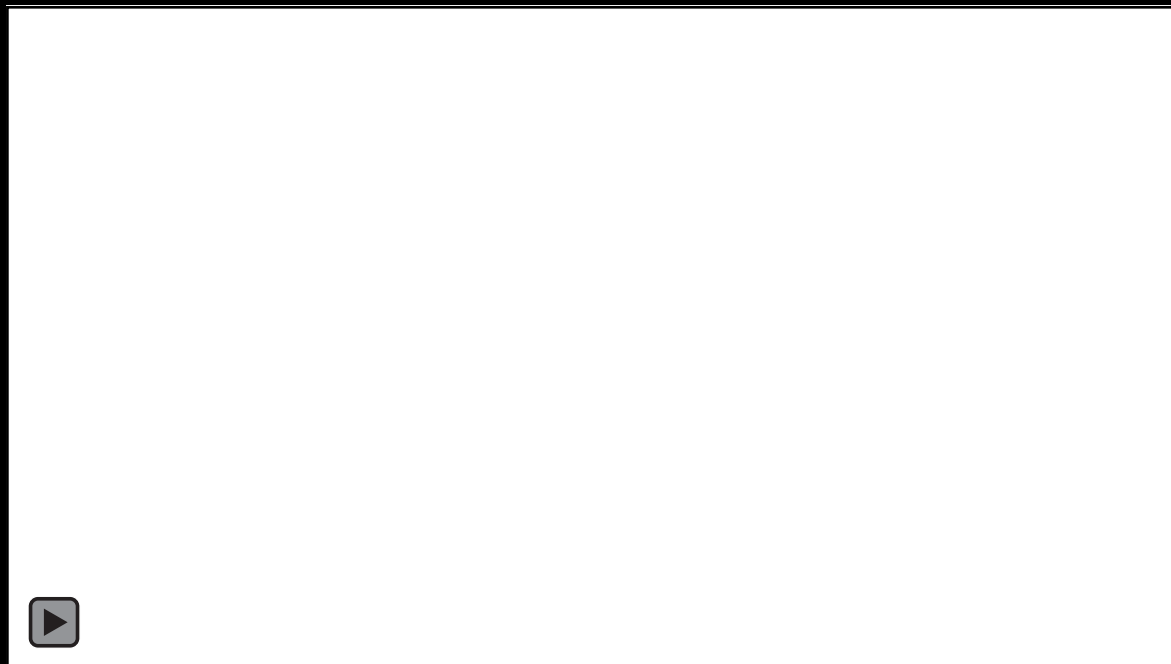
L-Band System

- Allows for transmission even in adverse weather

Satellite Crosslinks

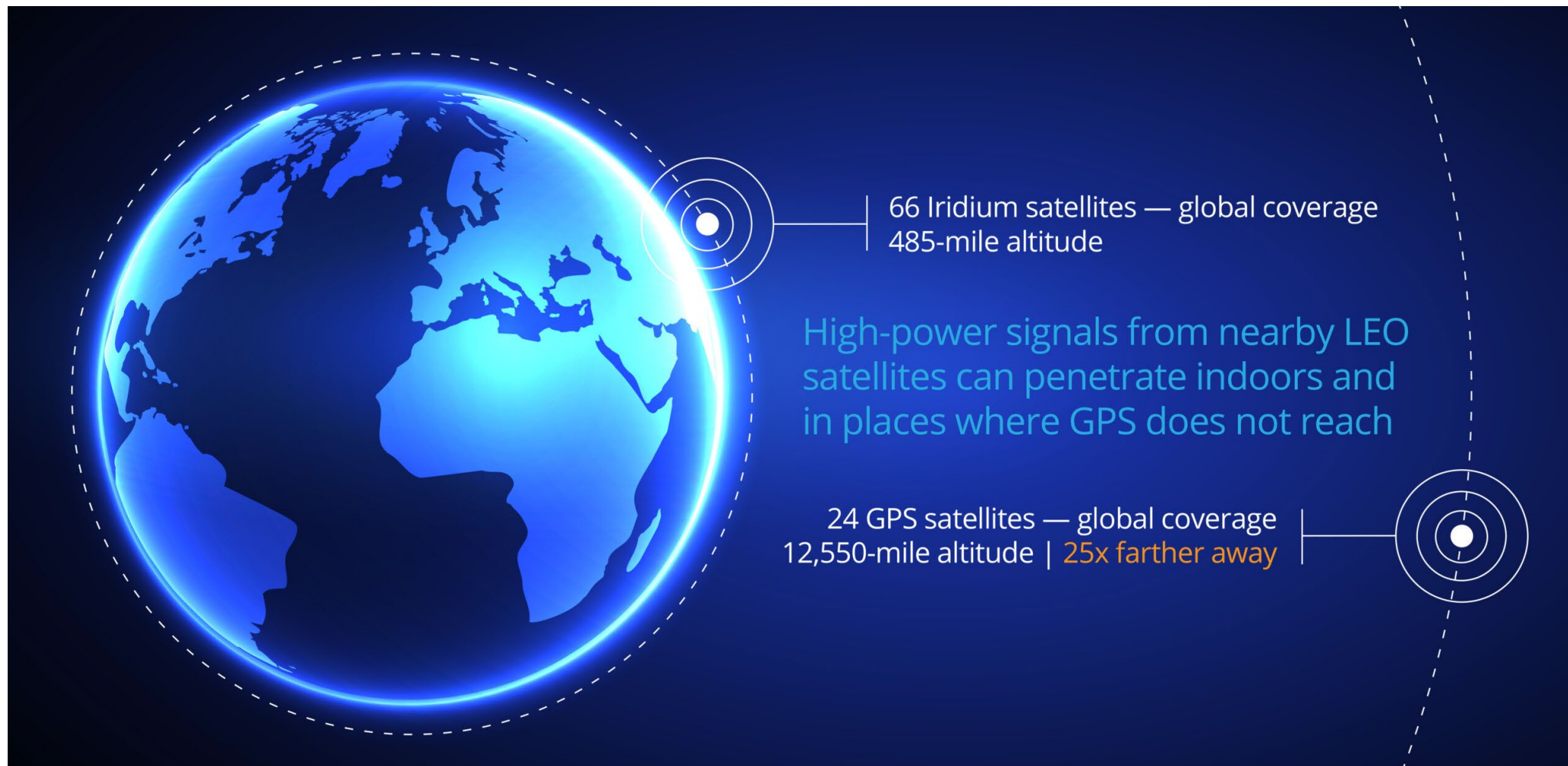
- Creates low-latency, resilient connections

Iridium's constellation provides truly global service



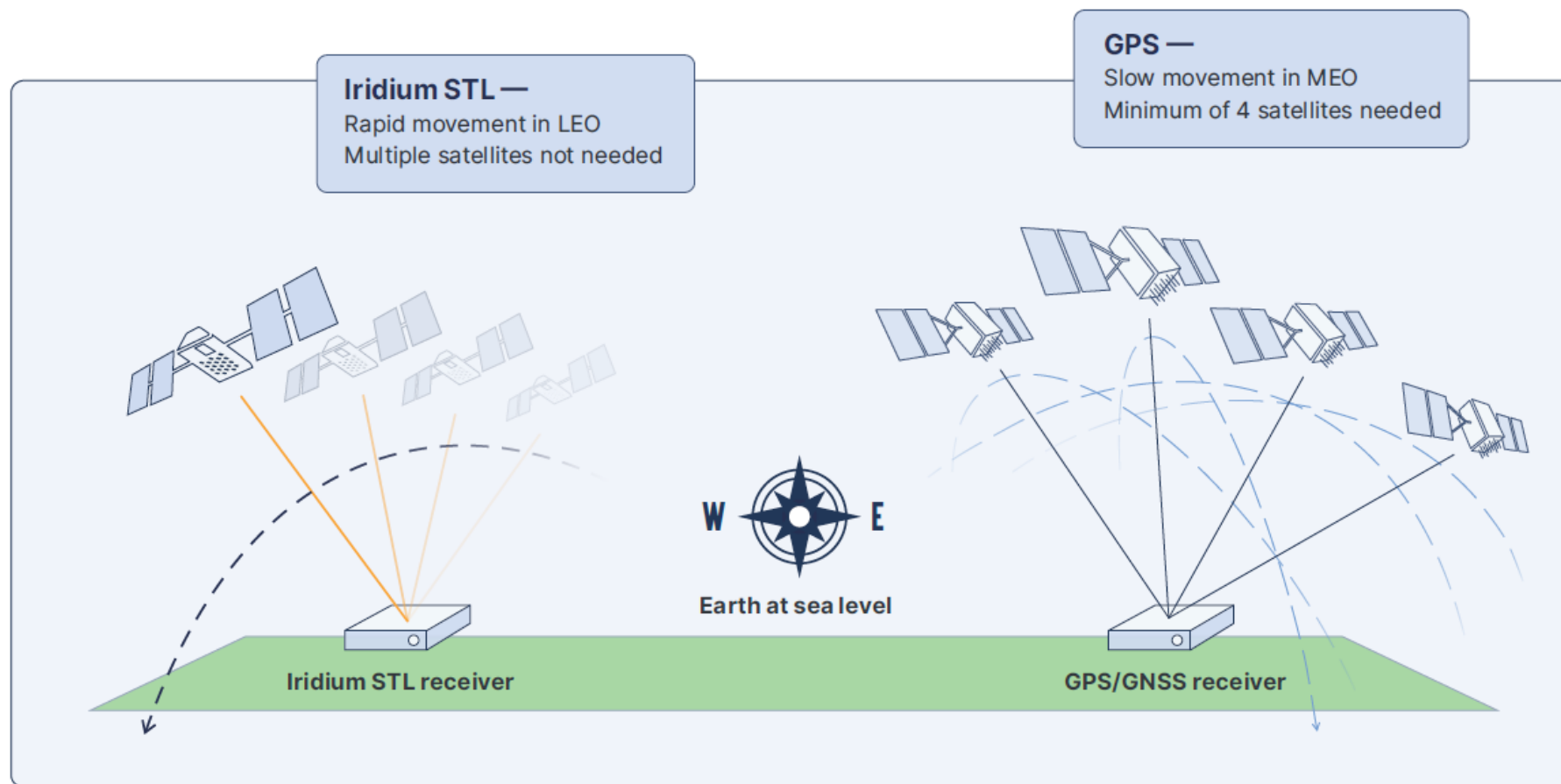


High Signal Power Relative to MEO / GEO





LEOs Are Fast Movers



Iridium (LEO) satellites circle the Earth **every 100 minutes**.

They move so fast that their ranging angle can change by up to 1 degree every 4 seconds, determining the user's location with **only 1 satellite in view**.

GPS/GNSS (MEO) satellites circle the Earth **every 12 hours**.

They move so slowly that **at least 4 satellites must be used** to determine the user's location.



Resilience to Direct Physical Attack

LEOs are less vulnerable to direct physical attacks than ground-based systems.





Surface Charging Effects

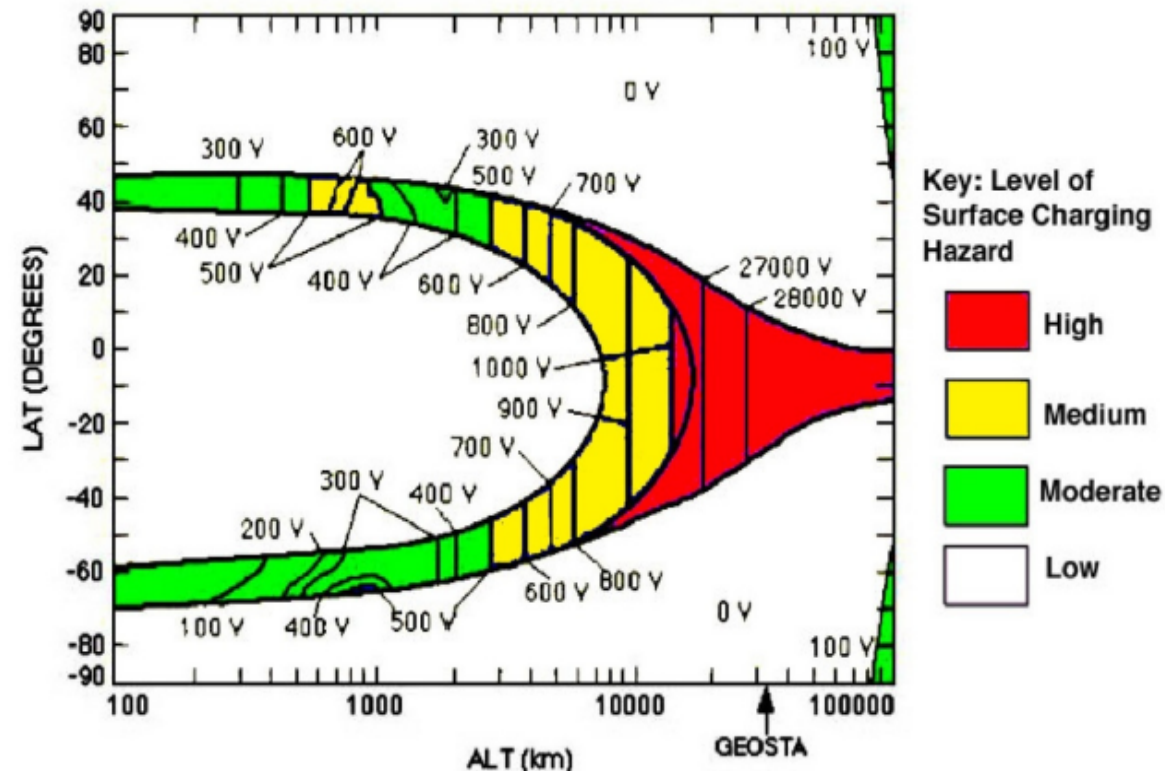


Figure 1—Earth Regimes of Concern for On-Orbit Surface Charging Hazards for Spacecraft Passing Through Indicated Latitude and Altitude (Evans and others (1989))

Chart source: NASA Technical Handbook, *Mitigating In-Space Charging Effects — A Guideline* (October 19, 2017)

https://ccmc.gsfc.nasa.gov/RoR_WWW/SWREDI/2018/nasa-hdbk-4002a_revalidated.pdf

SPACE

GEO

altitude 22,236 miles*

Earth Regimes of
Concern for Surface
Charging Hazards for
Spacecraft Passing
Through Indicated
(approximate) Altitude

MEO (GPS)
altitude 12,550 miles*

LEO (Iridium)
altitude 485 miles*

User

EARTH

* altitudes are shown to scale



Internal Charging Effects

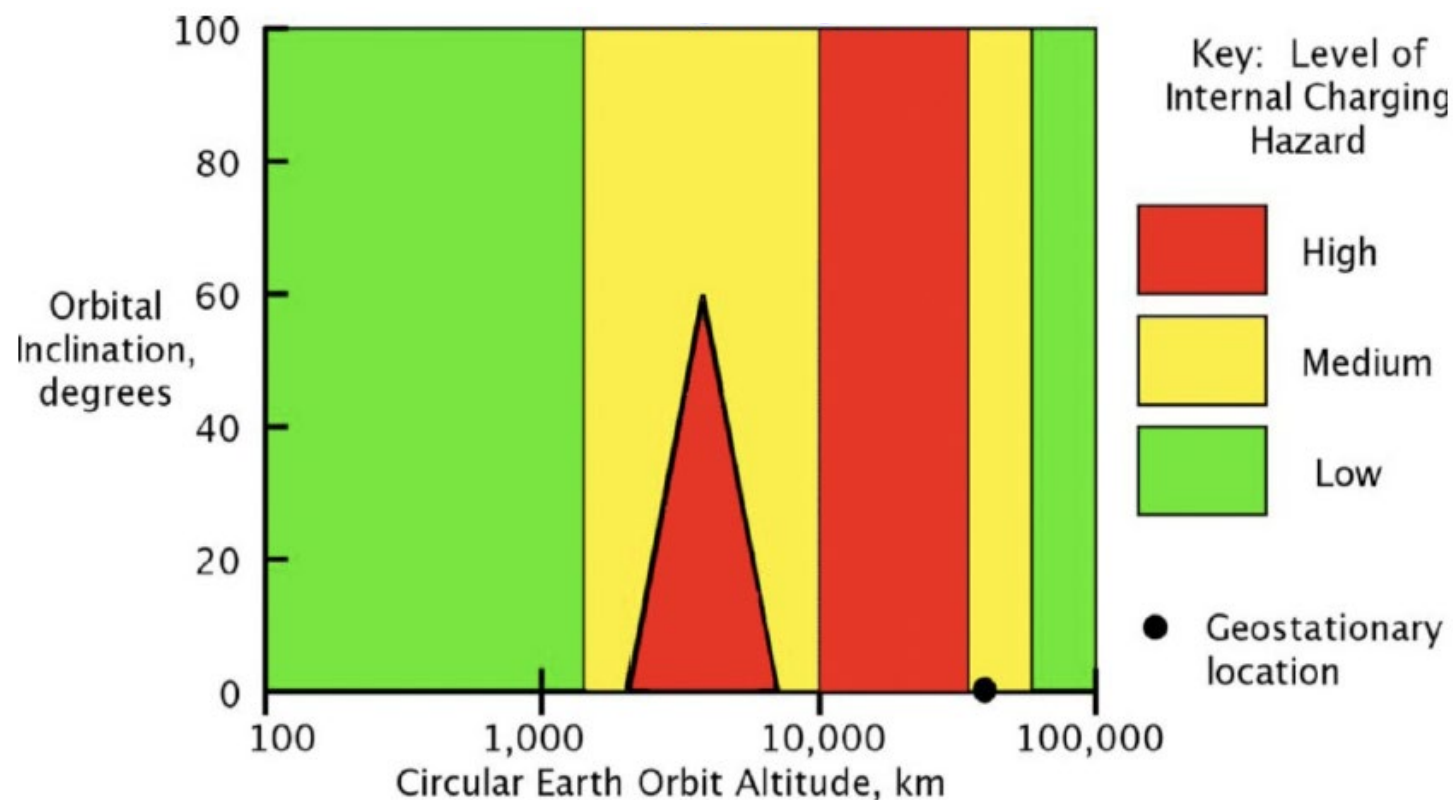


Figure 2—Earth Regimes of Concern for On-Orbit Internal Charging Hazards for Spacecraft with Circular Orbits

Chart source: NASA Technical Handbook, *Mitigating In-Space Charging Effects — A Guideline* (October 19, 2017)
https://ccmc.gsfc.nasa.gov/RoR_WWW/SWREDI/2018/nasa-hdbk-4002a_revalidated.pdf

SPACE

GEO

altitude 22,236 miles*

Concern for On-Orbit
Internal Charging
Hazards at
(approximate) altitude

MEO (GPS)
altitude 12,550 miles*

LEO (Iridium)
altitude 485 miles*

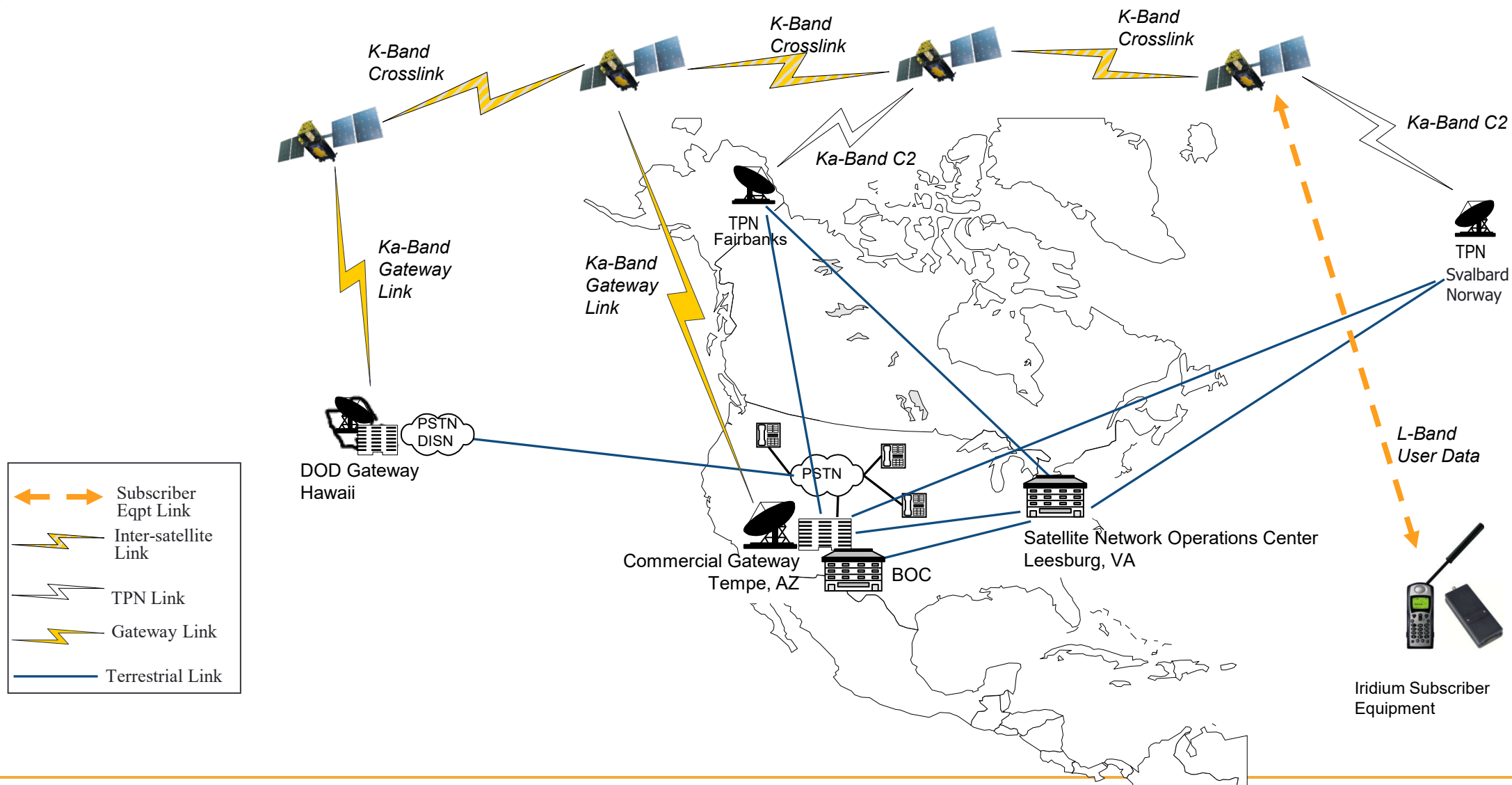
User

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* altitudes are shown to scale

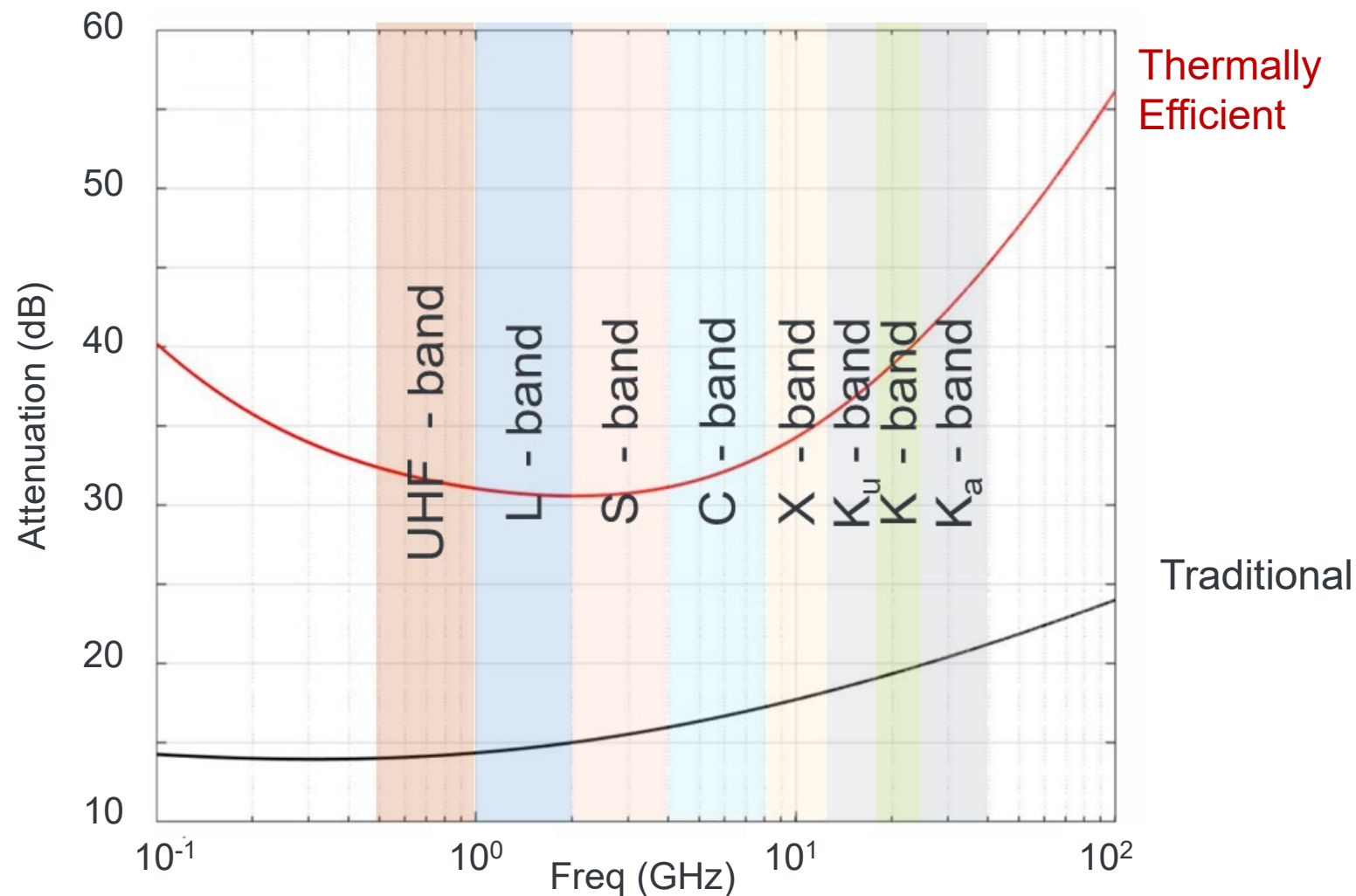


RF Frequency/Spectrum





ITU Analysis – Mean Building Entry Loss



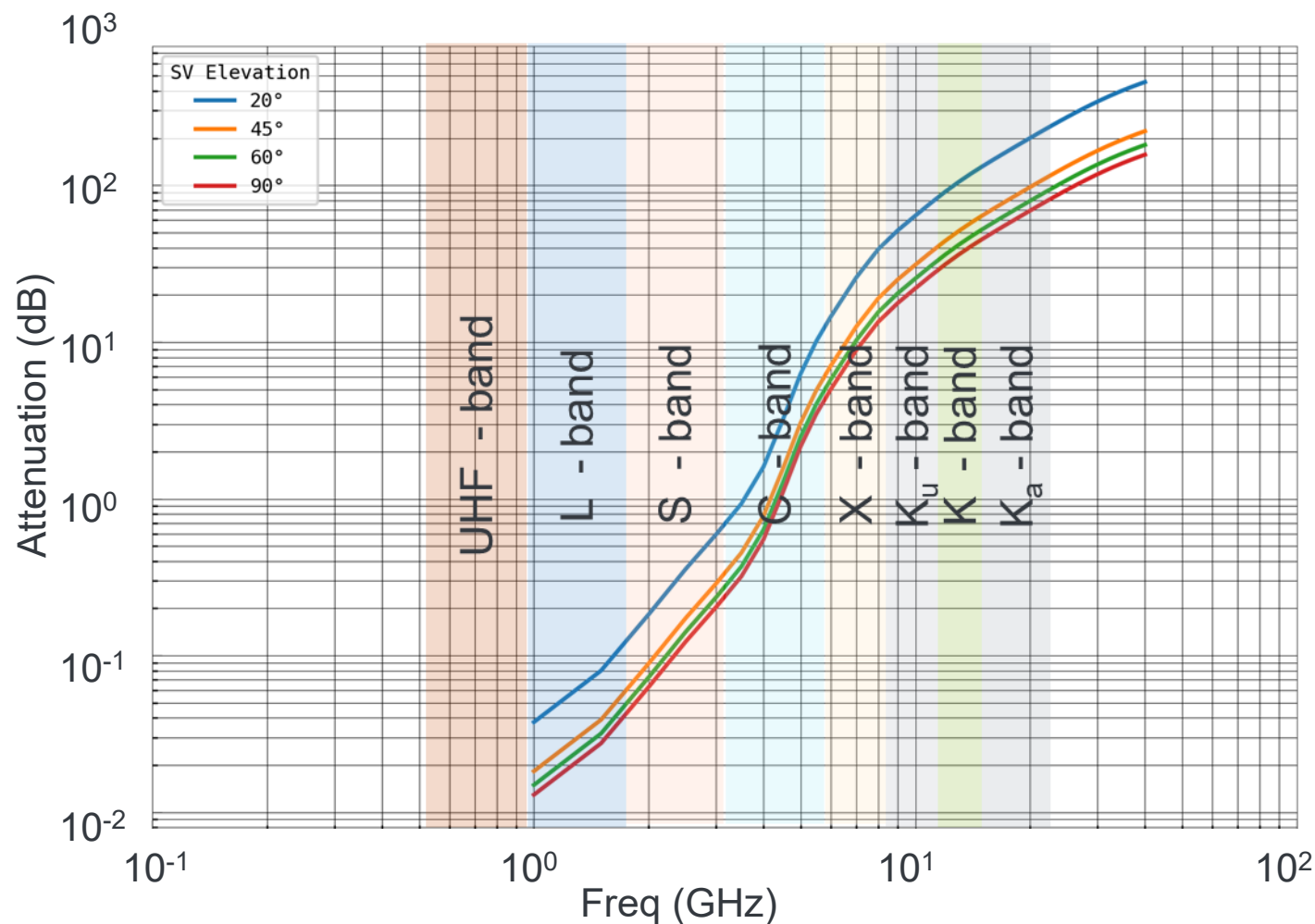
Model was developed with empirical data from ITU-R P.2346, and ITU-R P.2040

Traditional vs Modern building construction

“Typically, the presence of metallised glass windows, insulated cavity walls, thick reinforced concrete and metal foil back cladding is a good indication of a thermally-efficient building.”



ITU Analysis – Signal Attenuation During Thunderstorm



Propagation data and prediction methods required for the design of Earth-space telecommunication systems

Propagation losses

- Attenuation by rain, atmospheric gases
- Rain fade propagation loss plot developed using ITU ITU-R P.838-3 Model
- Conditions assumed to be thunderstorm in Florida



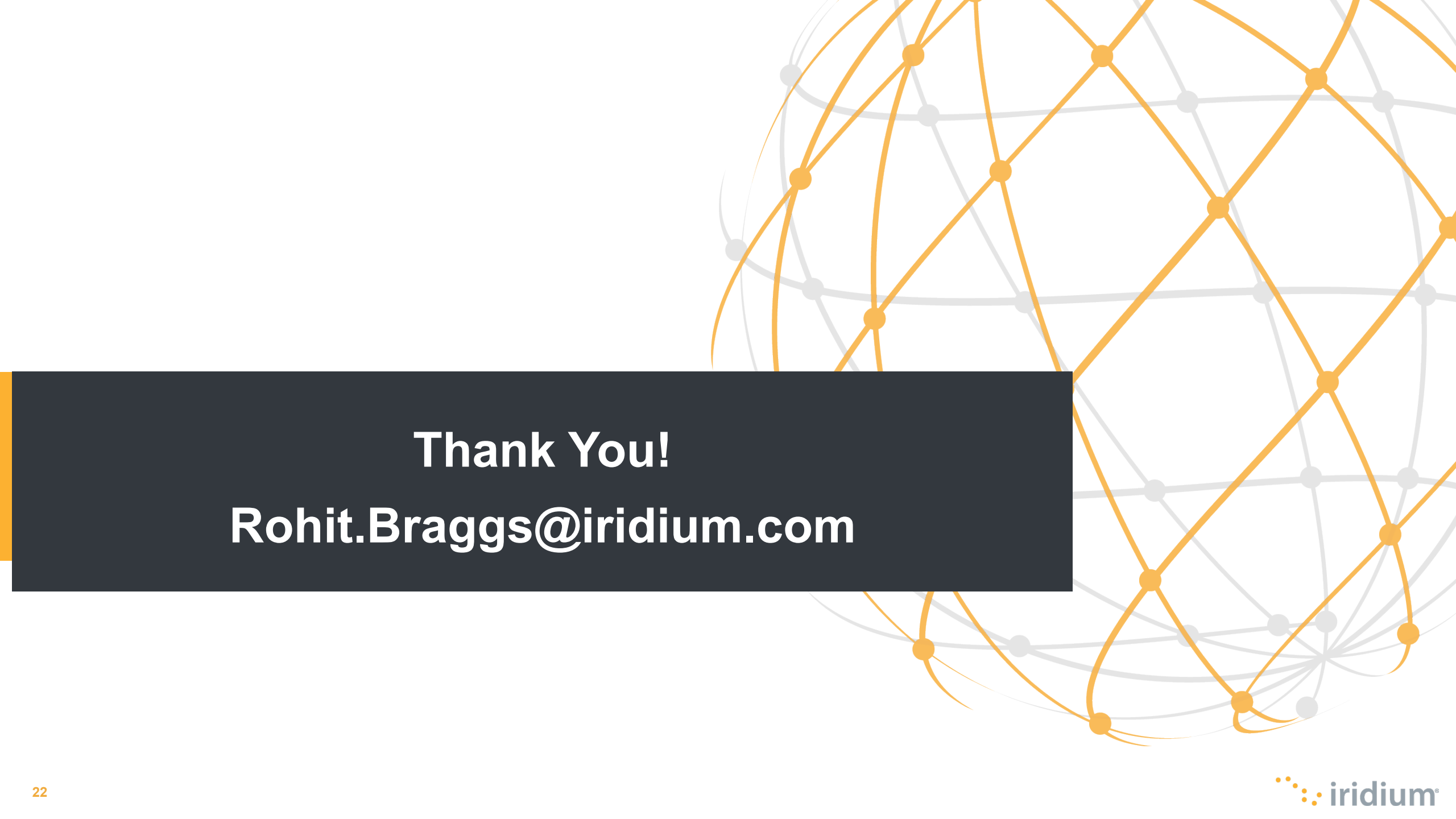
Bent Pipe versus Crosslinked Satellites

Crosslinks allow **communication** from anywhere to anywhere **without bent-pipe relays** around the globe.



Crosslinks also provide relative **timing measurements** in real-time, enabling calibration of the whole network from a **single ground site**.





Thank You!
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